HEAT TRANSFER

Introduction:

pefinition of heat transfer :-

-> Heat transfer may be defined as the transmission of energy from one region to another region as a result of temperature difference.

-> The study of heat transfer is carried out for the following purpose.

1. To estimate the rate of flow of energy as heat through the boundary of a system under study both study and transient conditions.

2. To determine the temperature field under steady and transient conditions.

> The areas covered under the discipline of heat

· Design of thermal and nuclear power plants including heat angines, steam generators, condensers and other heat exchange equipments.

· Internal combustion engines.

Refrigeration and air conditioning units.

· Design of cooling systems for electric motors, generators and transformers.

· thormal control

-> Difference's between thermodynamics and Heat

Thermodyramics

1. It deals with the equilibrium states of matter, and precludes the existence of a temperature gradient.

2. When a system changes from ore equilibrium state to another equilibrium state, thermodynamies helps to determine the quantity of work and heat interactions. It describes how much heat is to be exchanged during a process but does not himther how much the same could be achieved.

Heat transfer

1. It is a non equilibrium praces. (since temperature gradient must exist for exchange of heat to take place.

2. It helps to predict the distribution of temperature and to determine the rate at which energy is transferred across a surface of intrest due to temperature gradients at the surface, and difference of temperature between different surfaces.

-> Basic Laws of heat transfer: -

the following are the basic bows which govern heat transfer:

1. First low of thermodynamics: - when a system undergoes

when a system undergoes a thermodynamic cycle then the net heat supplied to the system from the sourroundings is equal to the network done by the system on its sourroundings.

System on its sourroundings.

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\$ da = f dw.

2. second how of thormodynamics:

Heat will flow automatically from one reservoir to another at a lower temperature, but not in apposite direction.

3. Low of conservation of mass: -

this law is used to determine the parameter of

4. Nowton's law of motion: -

these laws are used to determine fluid flow parameters.

5. The rate equations: -

these equations are made applicable depending upon the mode of heat transfer being considered.

-> Modes of Heat Transfer:-

Heat transfer which is defined as the transmission of energy from one region to another as a result of temperature difference takes place by the following three modes.

- i) conduction
- ii) convection
- iii) Radiation

1. conduction:

The transfer of heat between two solid bodies is called conduction. It depends on the difference in temperature of the hot and cold body. Example of conduction heat transfer is two bodies at different tom. perature kept in contact with each other. Another exam ple is heating one end of the metal like copper, due to conduction heat transfer the other end of the metal also gets heated pure conduction is only found in solids.

-> Fairier's law of heat conduction:

Fauier's law of heat conduction states as follows.

" the rate of flow of heat through a simple homogeneous solid is directly proportional to the area of the section of heat flow, and to change of temperature with respect to the length of the path of heat flow".

mathematically it ion be represented by

Qd A dt

Q=KAdt

where, a = heat flow through a body per unit time. A = surface area of heat flow, m2. dt = temperature difference in 10 or &. https://www.freshersnow.com/

dx = thickness of body in the direction of flow, m. 3

Ix = constant of proportionality and is known as thermal conductivity of the body.

Thus a = - k dt.

the -ve sign of k is to take care of the decreasing temperature along with the direction of increasing thickness or the direction of heat flow.

Assumptions:

- 1. conduction of heat takes place under stealy state conditions.
- 2. Temperature gradient is constant.
- 3. There is no internal heat generation.
- 4. the bounding surfaces are isothermal.
- 5. The material is homogeneous and isotropic.

some essential features of fourier's Low:

- 1. It is applicable to all matter.
- 2. It is based on experimental evidence and cannot be derived from first principle.
- 3. It is a vector expression.
- 4. It helps to define thermal conductivity

-> Thormal conductivity in

On = KA dt dx

K = Q dx https://www.freshersnow.com/ $K = Q \frac{dx}{dt}$ $K = Q \frac{dx}{dt}$ Q/A = Qwhere Q = heat flux

It is the amount of energy conducted through a body of unit area and unit thickness in unit time when the difference in temperature cause in the heat flow.

-> convection:

"convection is the transfer of heat within a fluid by mining of one portion of the fluid with another.

- 1. convection is possible only in a fluid medium and is directly linked with the transport of medium it self.
- 2. convection constitutes the macro-form of the heat transfer since macroscopic particles of a fluid moving in space cause the heat exchange.
- 3. the effectiveness of the convection depends upon the mining motion of the fluid.

-> Nawton's Low of cooling:

The rate equation for the convective heat transfer between a surface and an adjacent fluid is prescribed by Nowton's law of cooling.

a = hA(ts-tf)

a = Rate of conductive heat transfer https://www.freshersnow.com/ A = Area exposed to heat transfer

ts = surface temperature.

tr = fluid temperature,

h = co-efficient of convective heat transfer.

the units of h are,

h= Q = W or w/m2 or or W/m2 k

-> Radiationin

thex transfer afrom one body to another without transmitting medium is known as radiation. It is an electro magnetic wave phenomenon.

the transfer of heat by radiation occurs because hot body emits more heat than its receives and a cold body receives more heat than it emits. Radiant energy requires no medium for propagation and will pare through vaccum.

hows of Radiation: -

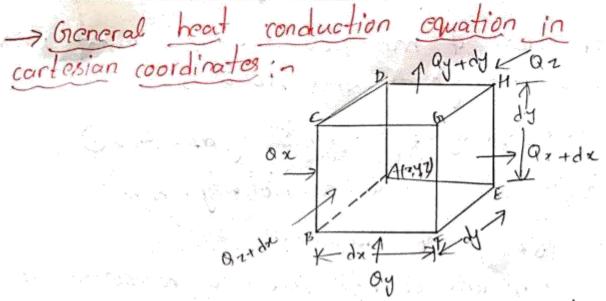
1. Wien's law: - It states that the wave length I'm corresponding to the maximum energy is inversely proportional to the absolute temperature of the hot body.

i.e \(\text{\max} \) \(\text{\text{T}} \) \(\text{\text{Constant}} \).

2. Kirchboff's low: - It states that the emissivity of the body at a particular temperature is numerically equal to its absorptivity for radiant energy from body at the same temperature.

3. The stefan-Boltzmann low: - the low states that the emissive power of a black body is directly proportional to fourth power of its absolute temperature.

i.e. \(\text{a} \text{2} \) \(\text{Prob} \) \(\text{1} \) \(\text{Prob} \) \(\text{1} \) \(\text{Prob} \) \(\text{1} \) \(\text{Prob} \) \(\text{2} \) \(\text{Prob} \) \(\text{1} \) \(\text{1} \) \(\text{Prob} \) \(\text{1} \) \(\text{1} \)



consider a small rectangular element of sides dx, dy, dz as shown in fig.

The energy balance of this rectangular element is obtained from first law of thermodynamics.

-> Net heat conducted into element from all the coordinate directions: -

Let 9x be the hoat flux in a direction of face ABCD and 9x+dx be the heat flux in a direction of face EFGH.

the rate of heat flow into the element in x direction through the face ABCD is

Gx = qx dydz = - kx ot dydz. -> 2 where k - + thormal conductivity, W/mk.

ST - temp gradiant.

The rate of heat flow into the element in x direction through the face ABCD is EFGH is

 $Q_x = q_x \frac{dy}{dz} = -kx \frac{\delta T}{\delta x} \frac{dy}{dz}$. $Q_x + dz = Q_x + \frac{\delta}{\delta x} (Q_x) \frac{\delta T}{dx}$

= - Kx &T dy dz + & [-kx &T dydz] dx

Qx+dx = - 12x of dydz - 18 [kx of] dxdydz -> 3

subtracting 2-3.

Qz - Qx + dx) = -kx &T dydz - [-kz &T dydz -

&x [kx ot] dz dy dz

= - Kx ST dydz + Kx Sx dydz + & [kx ST] dadydz

=> Qx - Qx+dx) = &x (kx ot) dx dy dz. -> (4) similarly,

$$Qy - Q(y + dy) = \frac{\delta}{\delta y} \left[ky \frac{\delta T}{\delta y} \right] dz dy dz - S$$

$$Qz \cdot Q(z + dz) = \frac{\delta}{\delta z} \left[k_z \frac{\delta T}{\delta z} \right] dx dy dz - S$$
Adding $(A) + (B) + (B)$

Not heart conducted = $\frac{\delta}{\delta x} \left(k_x \frac{\delta T}{\delta x} \right) dx dy dz + \frac{\delta}{\delta y} \left(k_y \frac{\delta T}{\delta y} \right) dx dy dz + \frac{\delta}{\delta z} \left(k_z \frac{\delta T}{\delta z} \right) dx dy dz.$

$$= \left[\frac{\delta}{\delta x} \left(k_x \frac{\delta T}{\delta x} \right) + \frac{\delta}{\delta y} \left(k_y \frac{\delta T}{\delta y} \right) + \frac{\delta}{\delta z} \left(k_z \frac{\delta T}{\delta z} \right) \right] dx dy dz.$$

$$= \frac{\delta}{\delta x} \left(k_x \frac{\delta T}{\delta x} \right) + \frac{\delta}{\delta y} \left(k_y \frac{\delta T}{\delta y} \right) + \frac{\delta}{\delta z} \left(k_z \frac{\delta T}{\delta z} \right) dx dy dz.$$
Heat should in the element:
$$= \frac{\delta}{\delta x} \left(k_x \frac{\delta T}{\delta x} \right) + \frac{\delta}{\delta y} \left(k_y \frac{\delta T}{\delta y} \right) + \frac{\delta}{\delta z} \left(k_z \frac{\delta T}{\delta z} \right) dx dy dz.$$

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$$= \frac{\delta}{\delta x} \left(k_x \frac{\delta T}{\delta x} \right) + \frac{\delta}{\delta z} \left(k_z \frac{\delta T}{\delta z} \right) dx dy dz.$$

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$$= \frac{\delta}{\delta x} \left(k_x \frac{\delta T}{\delta x} \right) dx dy dz.$$

$$= \frac{\delta$$

= [x(dx dy dz) x cpx &1 ot. - @ Heat generated within the clement: Heat generated with in the element is given by a = q dx dy dz.

$$= \int_{\delta n}^{\infty} \left[k_{x} \frac{\partial T}{\partial x} \right] + \frac{\partial}{\partial y} \left[k_{y} \frac{\partial T}{\partial y} \right] + \frac{\partial}{\partial z} \left[k_{z} \frac{\partial T}{\partial z} \right] \int_{\delta n}^{\infty} dx \, dy \, dz$$

$$+ q \, dx \, dy \, dz = \int_{\delta n}^{\infty} c_{p} \frac{\partial T}{\partial t} \, dx \, dy \, dz.$$

considering the material is isotropicat.

so, kx = ky = kz = k = constant.

$$\left\{\frac{\delta^{2}T}{\delta^{2}n^{2}} + \frac{\delta^{2}T}{\delta y^{2}} + \frac{\delta^{2}T}{\delta^{2}z^{2}}\right\} + q' = \int \varphi \frac{\delta T}{\delta L}$$

pivided by k

$$\frac{\delta^2 T}{\delta x^2} + \frac{\delta^2 T}{\delta y^2} + \frac{\delta^2 T}{\delta z^2} + \frac{q'}{K} = \frac{\rho \, \mathcal{E} \rho}{K} \, \frac{\delta T}{\delta t}.$$

$$\frac{\delta^2 T}{\delta x^2} + \frac{\delta^2 T}{\delta y^2} + \frac{\delta^2 T}{\delta z^2} + \frac{4}{k} = \frac{1}{\lambda} \cdot \frac{\delta T}{\delta t} - \emptyset$$

It is a general three dimentional heat conduction equation in cartesian coordinates.

thermal diffusivity is nothing but how fast heat is diffused through a material during changes of temperature builds hywhimes hersnow.com/

case 1):- No heat sources

In the absence of internal heat goneration, the equ @ reduces to

$$\frac{\partial^2 T}{\partial x^2} + \frac{\partial^2 T}{\partial y^2} + \frac{\partial^2 T}{\partial z^2} = \frac{1}{d} \frac{\partial T}{\partial t} - 0$$

This equation is known as diffusion equation.

cor) Fourier's equation.

case 2): - steady state conditions

In steady state conditions, the temperature does not change with time. So $\frac{\delta T}{\delta t} = 0$. The heat conduction equation reduces to

$$\frac{\partial^2 f}{\partial x^2} + \frac{\partial^2 f}{\partial y^2} + \frac{\partial^2 f}{\partial z^2} + \frac{\partial^2 f}{\partial z} = 0 \quad - \boxed{2}$$

$$cor)$$

This equation is known as poisson's equation.

In the absence of Internal heat generation equation becomes.

This equation is known as Laplace equation. https://www.freshersnow.com/

case iii): - one dimensional steady state heat conduction: If the temperature varies only in the x-direction the equation reduces to Bromes $\frac{3^2T}{8\pi^2} + \frac{9^2}{k} = 0$. — (B)

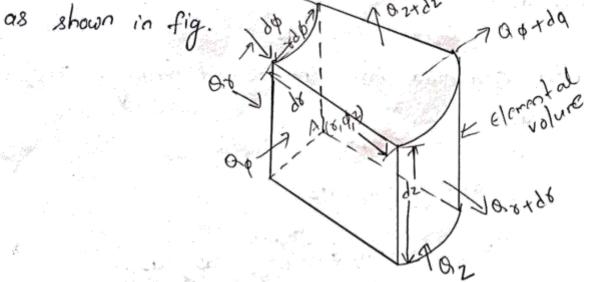
becomes $\frac{3^2T}{8\pi^2} = 0$. — (B) rase iv): - Two dimensional strady state heat conduction:
If the temperature varies only in the or and y

directions, the equations 10 becomes. In the absence of internal heat generation, equation (6) case v): n unsteady state, one dimensional, without internal heat generation: In unstrady state, the temperature changes with time, i'e st +0. so the general conduction equation 10 reduces to $\frac{87}{3n^2} = \frac{1}{\alpha} \frac{87}{88}$.

-> General heat conduction equation in cylinderal.

The general heat conduction equication in cartesian coordinates derived in the previous section is used for solids with rectangular boundaries like squares, aubes, slabes etc. But the cartesian coordinate system is not applicable for solids like aylinders, cones, spheres etc. For aylindrical solids, a cylindrical coordinate system is used.

consider an elemental volume having the coordinates (8, 0, 2) for three dimensional heat conduction analysis. as shown in fig. 89.01 A axia



the volume of the element $dv = 8 d\phi d8 dz$.

Let us assume that thermal conductivity (k), spatific heat (cp), and donsity (f) are constant.

The energy balance of this cylindrical element is obtained from first law of thermodynamics.

(Net heat conducted into element from all the co-ordinard directions)

Net heat conducted into element from all the co-ordinard directions:

Net heat conducted into element from all the co-ordinard directions:

Heat entering in the element through
$$(8, \emptyset)$$
 plane in time $d\theta$.

 $Q_z = -k (8 d\phi d8) \frac{\delta T}{\delta Z} d\theta$.

Heat leaving from the element through $(8, \emptyset)$ plane in time $d\theta$.

 $Q_z + \frac{\delta}{\delta Z} (Q_z) dz$.

Net heat conducted into the element through $(8, \emptyset)$ plane in time $d\theta$.

 $Q_z + \frac{\delta}{\delta Z} (Q_z) dz$.

Net heat conducted into the element through $(8, \emptyset)$ plane in time $d\theta$.

 $Q_z - Q_z + dz$
 $= -\frac{\delta}{\delta Z} (k (8 d\phi d8) (\frac{\delta T}{\delta Z}) d\theta$) dz .

 $= \frac{\delta}{\delta Z} (k (8 d\phi d8) (\frac{\delta T}{\delta Z}) d\theta$) dz .

Net heat conducted through $(8, \emptyset)$ plane $= k(\frac{\delta T}{\delta Z^2}) (d8 \cdot 8 d\phi dz)$ $d\theta$.

Net heat conducted through $(8, \emptyset)$ plane $= k(\frac{\delta T}{\delta Z^2}) (d8 \cdot 8 d\phi dz)$ $d\theta$.

as = -K (8 dødz) &T do Heat leaving from the element through (d, z) plane in time do

0x+dx = Qx+ 8 (Qx) dx. Not heat conducted into the element through (\$, 2) plane in time do.

= Qx - Qx+dx.

= - 3 (Ox) dx.

= - d (- k (8 dødz) (dt) de) dr.

= K (dx dødz). \frac{9}{9} (x \frac{91}{91}) do.

= K(dxxdp.dz) [3 + + + + 3 T] do.

Not heat conducted

= K(d8.8dp.dz)[37 + 1 27 do. through (p, z) plane

Heat entering in the element through (Z,8) plane in time do.

Qp = - K (d8.d2) ST d0.

Heat leaving from the element through (2,8) plane in time

Q+dq = Q+ + 8 (Qq) 8 dq.

Net heart conducted into the elament through (2,8) plane in time do.

$$Q_{\phi} - Q_{\phi} + d\phi = -\frac{\delta}{\delta \delta \phi} (Q_{\phi}) \delta d\phi.$$

$$= -\frac{\delta}{\delta \delta \phi} \left[-k \left(d \delta d z \right) \cdot \frac{\delta T}{\delta \delta \phi} d \theta \right] \delta d\phi.$$

$$= k \frac{\delta}{\delta \phi} \left[\frac{1}{\delta} \frac{\delta T}{\delta \phi} \right] (d \delta d \phi d z) d \theta.$$

$$= k \left[\frac{1}{\delta z} \frac{\delta^2 T}{\delta \phi^2} \right] (d \delta \delta \delta \phi d z) d \theta.$$

Net heat conducted

through (z, x) plane = K[\frac{1}{82} \frac{37}{392}] (dx xdp dz) d0.

-> Net heat conducted into element from all the

$$\left[\frac{\delta^{\frac{2}{4}}}{\delta^{\frac{2}{4}}} + \frac{1}{4} \frac{\delta^{\frac{2}{4}}}{\delta^{\frac{2}{4}}}\right] d\theta + k \left[\frac{8}{4} \frac{\delta^{\frac{2}{4}}}{\delta^{\frac{2}{4}}}\right] (d8 8 d\phi dz) d\theta.$$

Total heat generated within the element is given by

-> Heat stored in the element:n

The increase in internal energy of the element is caucil to the https://www.treshershow.com/ the demost.

Increase in internal energy

= net heat stored in the element

= (dx xd p dz) q \frac{\delta T}{\delta \theta} \times d\theta \to \frac{\delta T}{\delta \theta} \times d\theta \to \frac{\delta T}{\delta \theta} \times d\theta \to \frac{\delta T}{\delta \theta} \tag{6}

$$K(d88d\phi dz) d\theta \left[\frac{\partial^2 T}{\partial 8^2} + \frac{1}{8} \frac{\partial T}{\partial 8} + \frac{1}{8^2} \frac{\partial^2 T}{\partial \phi^2} + \frac{\partial^2 T}{\partial z^2} \right] +$$

q (dx xdødz)do = (cdx xdødz) cp ot xdo.

Divided by (dx xdpdz)do

$$\Rightarrow k \left(\frac{\delta^2 T}{\delta 8^2} + \frac{1}{5} \frac{\delta T}{\delta 8} + \frac{1}{5} \frac{\delta T}{\delta 8} + \frac{1}{5} \frac{\delta^2 T}{\delta 9^2} + \frac{\delta^2 T}{\delta 2^2} \right) + 4 = 6 cb \frac{21}{50}.$$

It is a general three dimensional heat conduction equation in cylindrical coordinates.

$$\frac{\delta_{1}^{2}}{\delta_{8}^{2}} + \frac{1}{8} \frac{\delta_{1}}{\delta_{8}} + \frac{1}{8^{2}} \frac{\delta_{1}^{2}}{\delta_{p}^{2}} + \frac{\delta_{1}^{2}}{\delta_{p}^{2}} + \frac{9}{67} + \frac{9}{4} = \frac{1}{4} \frac{\delta_{1}}{\delta_{0}}.$$

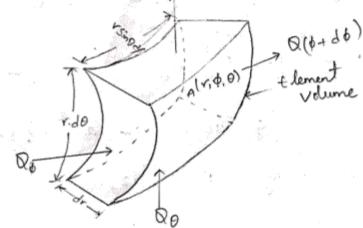
If the flow is steady, one dimensional $\frac{1}{pq}$.

and no heat generation, then the equation becomes

-> General heat conduction Equation in spherical mordinates. >

consider an elemental volume having the coordinates (8, 0, 0) for three dimensional heat conduction analysis.

as shown in fig.



The volume of the doment = dx xd0 xsin 0 dg.

-> Not heat conducted into element: Heat entering in the element through (x, 8) plane in thre to p direction.

Qb = -k(dx xq0) = of qo. treat leaving from the element through (8,0) plane in & direction.

$$Q \phi + d \phi = Q \phi + \frac{\delta}{\sqrt{5}} (Q \phi) d \phi$$
. $\sqrt{5} \sin \theta d \phi$. $\sqrt{2}$
https://www.freshersnow.com/

Net heat conducted into the element through (8,0) = Qp - Qp+dp = - 10 (00)dp. = - 2 [/kedx (8/10) \$5,40 \$ \$0 J.do = \$5/00 St = K (08/800. = - I sind & (Qp) Ksinddø. = K (do 8 do 8 sino do) 1 85 do do. Heat flow in 8-\$ plane, 0-direction. Qo = - K (dx xsing. dp) x dp do. 80+00 = 90 + 80 (00) 800 Heat conducted into the element through (8, 9)

 $= \frac{\partial}{\partial \theta} - \frac{\partial}{\partial \theta} = \frac{$

$$=\frac{k}{8}\frac{dx \cdot 8d8 \cdot 8d8}{8}\frac{d\theta}{8}\frac{\delta}{8}\left(\sin\theta \cdot \frac{\delta T}{\delta \theta}\right)d\theta$$

$$= 1c(d8 \times 8d8 \times 8in\theta \cdot 8d9)\frac{1}{8^{2}\sin\theta}\frac{\delta}{8\theta}\left(\sin\theta \frac{\delta T}{\delta \theta}\right)d\theta$$
Heat flow in θ -p plane, 8 -direction.
$$Q_{8} = -k(8d8 \times 8in\theta \cdot 8d9)\frac{\delta T}{\delta x}d\theta$$

$$Q_{8} + d_{8} = Q_{8} + \frac{\delta}{8}(Q_{8})d_{8}.$$
Not heat conducted in the element
$$= Q_{8} - Q_{8} + d_{8}.$$

$$= \frac{\delta}{\delta x}\left[-(K \times 8d8 \times 8in\theta \cdot 8d9)\frac{\delta T}{\delta x}\cdot d\theta\right]d\delta$$

$$= 1c(d8 \times 8d8 \cdot 8sin\theta \cdot dp)\frac{1}{8^{2}}\frac{\delta}{\delta x}\left(8^{2}\frac{\delta T}{\delta x}\right)d\theta$$

$$= 1c(d8 \times 8d8 \cdot 8sin8 \cdot dp)\frac{1}{8^{2}}\frac{\delta}{\delta x}\left(8^{2}\frac{\delta T}{\delta x}\right)d\theta$$
Not heat accumulated in the element
$$= k(d8 \times 8d8 \cdot 8sin8 \cdot dp)\left(\frac{1}{8^{2}sin9}\frac{\delta}{\delta y} + \frac{1}{8^{2}sin9}\right)\frac{\delta}{\delta \theta}\left(\sin\theta \frac{\delta T}{\delta \theta}\right) + \frac{1}{8^{2}}\frac{\delta}{\delta x}\left(8^{2}\frac{\delta T}{\delta x}\right)d\theta$$

$$\Rightarrow \text{Heat generated within the element: } 1$$
total heat generated within the element is given by $Q_{8} = Q_{8}(d8 \times d\theta \cdot 8sin\theta \cdot d\phi)d\theta$.

NOW,

(Cd8.8d0.85100.dp) q. 3T.do.

dividing both sides by k (dr. 8d0.85in 0dg) do.

$$\left[\frac{1}{8^2 \sin^2 \theta} \frac{\delta^2 T}{\delta \phi^2} + \frac{1}{8^2 \sin^2 \theta} \frac{\delta}{\delta \theta} \left(\sin \theta \frac{\delta}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \pi} \left(8^2 \frac{\delta T}{\delta \pi} \right) \right] + \frac{9}{16} \left[8^2 \frac{\delta T}{\delta \theta} \right] + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta T}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta T}{\delta \theta} \right) \right] + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta T}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta T}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta T}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta T}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta T}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta T}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta T}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta T}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta T}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta T}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta T}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta T}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta T}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta T}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta T}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta}{\delta \theta} \right) + \frac{1}{8^2} \frac{\delta}{\delta \theta} \left(8^2 \frac{\delta}{\delta \theta}$$

the above equation is the general heat conduction equation in spherical coordinates.

In case there are not heat sources present and the heat flow is steady and one dimensional, then the equation reduces to.

-> one dimensional steady state conduction heat. 13.

1. place walls and composite walls.

2. Hallow allinders and composite allinders.

3. spheres.

1. conduction of head through a slab on plane wall;

consider a slab of uniform thermal conductivity K, thickness L, with inner & temperature T, and outer temperature

Let us comider a small elemental area of thickness dx. from journar law of conduction, w. k.T

$$Q = -kA \frac{dT}{dx}$$

Integrating the above equation between the limits

of o to b, and T, and To

The general heat conduction equation in cartesian coordinates is given by.

If the conduction takes place under the conditions steady state $(\frac{\delta T}{\delta \theta} = 0)$, one dimensional $(\frac{\delta T}{\delta y^2} = \frac{\delta^2 T}{\delta z^2} = 0)$ and with no internal heat generation $(\frac{q}{k} = 0)$ than the above equation reduces to.

$$\frac{\partial^2 T}{\partial x^2} = 0. \quad \text{coij} \quad \frac{d^2 T}{dx^2} = 0 \quad -0$$

By integrating the above equation twice, we get.

$$\frac{dE}{dx} = C$$

c, & c2 are arbitary constants.

at
$$n=0$$
, $t=t_1$
at $n=h$ $t=t_2$

substitute these values than.

$$t_1 = 0 + c_2$$
 and $t_2 = c_1 b + c_2$

$$c_2 = t_1$$

$$c_1 = \frac{t_2 - c_2}{b}$$

$$c_1 = t_2 - t_1$$

$$T = (T_2 - T_1) \times + c_2$$

$$T = \left(\frac{T_2 - T_1}{L}\right) \times + C_2$$

$$T = \left(\frac{T_2 - T_1}{L}\right) \times + T_1$$

(12-T1) x + LJT1 = TL.

$$(T_2 \cdot T_1) x = Tb - bT_1$$

 $x = \frac{(T - T_1)b}{(T_2 - T_1)} = \frac{(T_1 - T_2)}{(T_1 - T_2)} b$.

conduction of heat through hollow cylinder: new radius 81, auter radius 82, inner tem T1, outer temp T2 and thermal conductivity k.

Let us consider a small elemental area of thickness "ds".

from fourier Law of conduction, W. K.T.

Area of the cylinder is 2718L.

$$A = 2\pi 8 h$$
, $Q = -kA = \frac{d}{d8}$
 $SO, Q = -k 2\pi 8 h = \frac{d}{d8}$

Integrating the above equations from 8, to 82 and Ti to To. 9 · 5 dx = - k2 TL 5 dT. Q (hx) = - K2TL (T)T2 Q(ln 82- ln 81) = -K2 TL (T2-T1) a ln (82) = 2Thk (T1-T2) Q = 2TKL (T, -72) In (82/81) Q = T1-72 27 KL In (82/81) Q = DTovorall DT = T, -T2, R= 1716 In (82/81). -> conduction of heat through Hollow sphere: consider a hollow sphere of inner radius & , outer radius & inter tem Ti, auter temperature 72 and thermal conductivity K. Let us consider a small elemental of thickness do from fourier law of heat conduction, W. K.T.

Area of sphere is
$$4\pi 8^2$$

Area of sphere is $4\pi 8^2$
 $A = 4\pi 8^2$.

 $A = 4\pi 8$

-> Heat conduction through a composite wall:n

consider the transmission of heat through a composite wall consisting of a number of slabs.

Let WA, UB, be = Thickness of slabs A, B, C

KA, KB Kc = thormal conductivities of the slabs

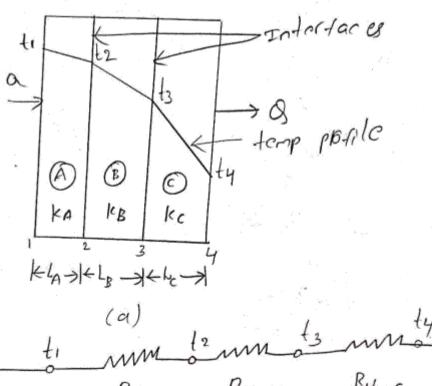
ti, ty (t,7ty) = temp at the wall surfaces landy

to, to = temp at the innerfaces 2 and 3.

since the quantity of heat transmitted per unit time through each slab/layer is some,

Q = KAA. (t1-t2) = KB. A (t2-t3) = Kc A (t3-t4)

Assuming that there is a perfect contact between the layers and no temp drop occurs are the interface between the materials.



Rearranging the above equation we get,
$$\frac{1}{1-12} = \frac{Q L_A}{k_A A} \qquad 0$$

$$\frac{1}{12-13} = \frac{Q L_B}{k_B A} \qquad 0$$

$$\frac{1}{12-13} = \frac{Q L_B}{k_B A} \qquad 0$$

$$\frac{1}{12-13} = \frac{Q L_B}{k_B A} \qquad 0$$
Adding 1,2,3 equations.
$$\frac{1}{1-13} = \frac{Q L_A}{k_A A} + \frac{L_B}{k_A A} + \frac{L_B}{k_A A} \qquad K_A$$

dding 1,2,3 Equations.

$$1, -ty = Q \left(\frac{b_A}{k_B A} + \frac{b_B}{k_B B} + \frac{b_C}{k_C A} \right)$$

$$Q = \frac{A C t_1 - t_U}{\left(\frac{L_A}{K_A} + \frac{L_B}{I_B} + \frac{L_C}{I_C}\right)}$$

If the composite wall consists of n slabs/ layers. then , Q = (+1-+(n+1)) Q = DToverall

In order to solve more complen problems involving both series and parallel thermal resistances, the electrical analogy may be used.

Thermal contact resistance: 7

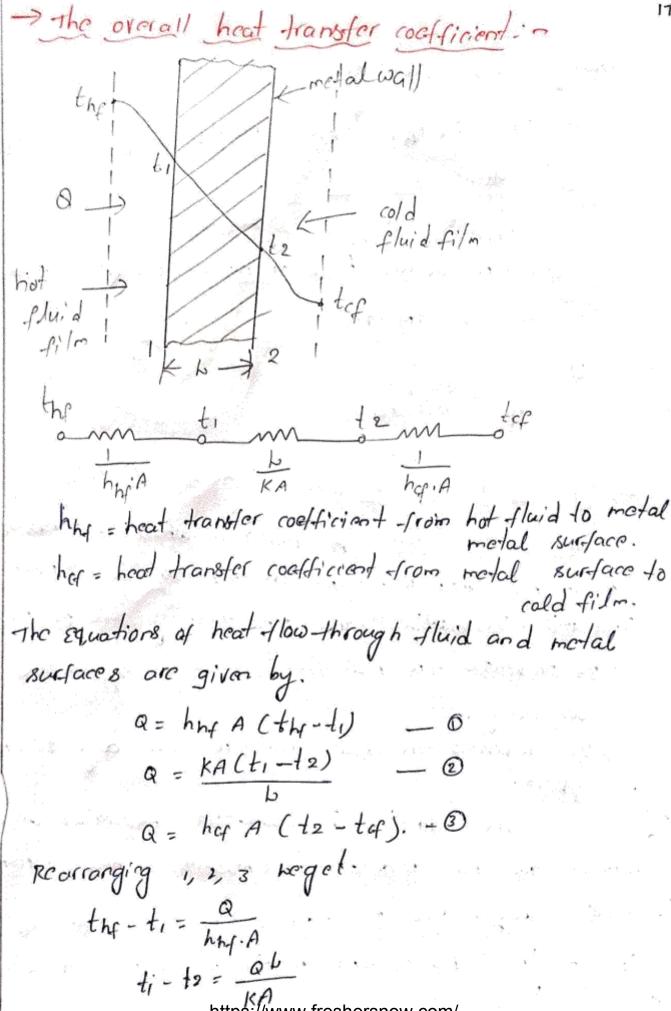
In composite wall, the calculations of heat-flow are made on the assumptions.

i) the contact between the adjacent layers is prefect.

ii) At the interface there is no fall of temperature.

iii) At the interface the temp is continuous,

In real systems, however due to surface roughness and spaces (filled with airs the contact surfaces touch only at discrete locations. That the area available for the flow of heat at the interface will be small compared to geometric face area. Due to this reduced area and presence of air, a large resistance to heat flow at the interface occurs. This resistance is known as thermal contact resistance.



the top =
$$\frac{a}{h_{ce}A}$$
.

the top = $a(\frac{1}{h_{ce}A} + \frac{b}{kA} + \frac{1}{h_{ce}A})$
 $a = \frac{a(t_{H} - t_{ce})}{\frac{1}{h_{H}A} + \frac{b}{k} + \frac{1}{h_{ce}A}}$

If v is the overall heat transfer coefficient, then

 $a = vA(t_{H} - t_{ce}A)$
 $v = \frac{1}{t_{H}A} + \frac{b}{k} + \frac{1}{h_{ce}A}$

-> critical radius of insulation:

Insulation definition in A material which retails the flow of heat with reasonable effectiveness is known as Insulation. Insulation serves the following two purposes:

i) It prevents the heat flow from the system to surroundings.

ii) It prevents the heat from the surrounding to the

the addition of insulation material on a surface does not reduce the amount of heat transfer rate always. Infact under certain circumstances it actually increases the heat loss up to certain

thickness of insulation. The radius of insulation for which the heat transfer is maximum is called critical radius of insulation and the corresponding thickness is called critical thickness. If the thickness is further increased, the heat loss will be reduced.

-> critical radius of insulation for a cylindering mulation

consider a solid cylinder _ _ ! of radius x, insulated with an insulation of -1 hickness (82-81) as shown in fig.

het b= length of cylinder ti = surface ten of cylinder

tair = ten of air.

ho: heart transfer coefficient at the arter surface of the insulation.

K = thermal conductivity of insulating material. the rate of heat transfer from the surface of the solid cylinder to the surroundings is given by.

a = 2Th (t, -tair) In (82/81) + 1

from the above equ that as & increases, the factor In (82 to) increases but the factor to be decreases.

thus a becomes maximum when the denominator become minimum the required condition is

$$\frac{d}{ds_2} \left(\frac{\ln(s_2/s_1)}{K} + \frac{1}{h_0 s_2} \right) = 0.$$

$$\frac{1}{K} \frac{1}{s_2} + \frac{1}{h_0} \left(\frac{1}{s_2^2} \right) = 0.$$

$$\frac{1}{K} \frac{1}{s_2} = \frac{1}{h_0} \frac{1}{s_2^2}$$

$$\frac{1}{K} = \frac{1}{h_0} \frac{1}{s_2^2}$$

pro: 1. calculate the critical radius of insulation for as bestos

[k = 0.172 W/m K) surrounding a pipe and exposed to room

air at 300 k with h = 2.8 W/m K. calculate the heat loss

from a 475 K, 60mm diameter pipe when covered with

the critical radius of insulation, and without insulation.

i) the critical radius of insulation,
$$8c = \frac{k}{ho} = \frac{0.172}{9.8} = 0.06143 \,\text{m} \quad (or) \quad 61.43 \,\text{mm}.$$

-> plane wall with uniform heat generation in

consider a plane wall of thickness is of uniform thormal conductivity K. Let the wall surfaces are maintained at temperatures to and to.

Let us assure that the heat flow 19 one dimensional

* steady https://www.freshersnow.com/

Heat conducted in ad a distance dx. Qx = - KA dt Heat generated in the element ag = Adxag. Heat conducted out at distance twa ON (xtdx) Quitdr = Qx+ dx (Qn) dx An energy balance on the element of thick dx is given by. ax+ ag = ax+dx Gx+0g = 0x + d (Qx) dx Og = dx (Ox).dx Adrag = drl-KA fr Jdx $= -KA \frac{d^2t}{dx^2} dx$ d2t + q, =0 →0 The first and second integration, of Equ O gives respectively. # = -9 x + 9. : t= -9 x2 + 4x + C2

using these boundary conditions in the above equation

$$c_2 = t = t\omega$$

substituting these values of G and cz in the above Equ.

In order to determine the bocation of the maximum temp, differentiating the Equ. w. 8.1. is and Equate to zero.

$$\alpha : \frac{b}{2}$$