Time: 3 hours

 $\mathbf{R05}$

Set No. 2

III B.Tech II Semester Examinations, APRIL 2011 COMPUTATIONAL AERODYNAMICS Aeronautical Engineering

Max Marks: 80

Answer any FIVE Questions All Questions carry equal marks *****

- 1. (a) What is Computational Fluid Dynamics? Explain with any two examples its applications in industrial manufacturing industry.
 - (b) Explain with diagrams the flow models that use finite control volume approach and infinitesimal fluid element approach. [8+8]
- 2. What are the available structured grid generation techniques and explain the conformal mapping method. [16]
- 3. (a) Why computational fluid dynamics makes a distinction between conservation and non-conservation forms of governing equations? Explain with examples.
 - (b) Explain why conservation form of governing equations is important for calculations using shock capturing method with the help of an example of flow across a normal shock wave. [8+8]
- 4. Prove that an expression for second-order central difference expression for the mixed derivative is $\left(\frac{\partial^2 u}{\partial x \partial y}\right)_{i,j} = \frac{u_{i+1,j+1} u_{i+1,j-1} u_{i-1,j+1} + u_{i-1,j-1}}{4\Delta x \Delta y} + O\left[\left(\Delta x\right)^2, \left(\Delta y\right)^2\right]$ can be obtained. [16]
- 5. (a) What is grid adaption
 - (b) Its need in a flow domain.
 - (c) Role played by it on the numerical solution with an example for two dimensional steady flow in a pipe. [4+4+8]
- 6. (a) Derive the continuity equation $\partial \rho / \partial t + \nabla \bullet (\rho V) = 0$ assuming appropriate flow model.
 - (b) Derive the continuity equation $D\rho/Dt + \rho\nabla \bullet \nabla = 0$ assuming appropriate flow model. [8+8]
- 7. Classify the following partial differential equations according to their nature as elliptic/parabolic or hyperbolic
 - (a) Unsteady Thermal Conduction Equation: $\partial T/\partial t = \alpha \partial^2 T/\partial t^2$
 - (b) Laplace's Equation: $\partial^2 \phi / \partial x^2 + \partial^2 \phi / \partial y^2 = 0$
 - (c) Second-order wave equation: $\partial^2 u / \partial t^2 = c^2 \partial^2 u / \partial x^2$
 - (d) First order wave equation: $\partial u/\partial t + c\partial u/\partial x = 0.$ [4+4+4]

 $\mathbf{R05}$

Set No. 2

8. What are metrics and derive the relationship between the direct and inverse metrics. [16]

i.e.
$$\frac{\partial \xi}{\partial x} = \frac{1}{J} \quad \frac{\partial y}{\partial \eta}$$
 $\frac{\partial \eta}{\partial x} = -\frac{1}{J} \quad \frac{\partial y}{\partial \xi}$
 $\frac{\partial \xi}{\partial y} = -\frac{1}{J} \quad \frac{\partial x}{\partial \eta}$ $\frac{\partial \eta}{\partial y} = \frac{1}{J} \quad \frac{\partial x}{\partial \xi}$

 $\mathbf{R05}$

Set No. 4

III B.Tech II Semester Examinations, APRIL 2011 COMPUTATIONAL AERODYNAMICS Aeronautical Engineering

Time: 3 hours

Max Marks: 80

Answer any FIVE Questions All Questions carry equal marks *****

- 1. Prove that an expression for second-order central difference expression for the mixed derivative is $\left(\frac{\partial^2 u}{\partial x \partial y}\right)_{i,j} = \frac{u_{i+1,j+1} u_{i+1,j-1} u_{i-1,j+1} + u_{i-1,j-1}}{4\Delta x \Delta y} + O\left[\left(\Delta x\right)^2, \left(\Delta y\right)^2\right]$ can be obtained. [16]
- 2. Classify the following partial differential equations according to their nature as elliptic/parabolic or hyperbolic
 - (a) Unsteady Thermal Conduction Equation: $\partial T/\partial t = \alpha \partial^2 T/\partial t^2$
 - (b) Laplace's Equation: $\partial^2 \phi / \partial x^2 + \partial^2 \phi / \partial y^2 = 0$
 - (c) Second-order wave equation: $\partial^2 u / \partial t^2 = c^2 \partial^2 u / \partial x^2$
 - (d) First order wave equation: $\partial u/\partial t + c\partial u/\partial x = 0.$ [4+4+4+4]
- 3. (a) Why computational fluid dynamics makes a distinction between conservation and non-conservation forms of governing equations? Explain with examples.
 - (b) Explain why conservation form of governing equations is important for calculations using shock capturing method with the help of an example of flow across a normal shock wave. [8+8]
- 4. (a) What is Computational Fluid Dynamics? Explain with any two examples its applications in industrial manufacturing industry.
 - (b) Explain with diagrams the flow models that use finite control volume approach and infinitesimal fluid element approach. [8+8]
- 5. (a) What is grid adaption
 - (b) Its need in a flow domain.
 - (c) Role played by it on the numerical solution with an example for two dimensional steady flow in a pipe. [4+4+8]
- 6. (a) Derive the continuity equation $\partial \rho / \partial t + \nabla \bullet (\rho V) = 0$ assuming appropriate flow model.
 - (b) Derive the continuity equation $D\rho/Dt + \rho\nabla \bullet \nabla = 0$ assuming appropriate flow model. [8+8]
- 7. What are the available structured grid generation techniques and explain the conformal mapping method. [16]

 $\mathbf{R05}$

Set No. 4

8. What are metrics and derive the relationship between the direct and inverse metrics. [16]

i.e.
$$\frac{\partial \xi}{\partial x} = \frac{1}{J} \quad \frac{\partial y}{\partial \eta}$$
 $\frac{\partial \eta}{\partial x} = -\frac{1}{J} \quad \frac{\partial y}{\partial \xi}$
 $\frac{\partial \xi}{\partial y} = -\frac{1}{J} \quad \frac{\partial x}{\partial \eta}$ $\frac{\partial \eta}{\partial y} = \frac{1}{J} \quad \frac{\partial x}{\partial \xi}$

Time: 3 hours

 $\mathbf{R05}$

Set No. 1

III B.Tech II Semester Examinations, APRIL 2011 COMPUTATIONAL AERODYNAMICS Aeronautical Engineering

Max Marks: 80

Answer any FIVE Questions All Questions carry equal marks *****

1. What are metrics and derive the relationship between the direct and inverse metrics. [16]

> *i.e.* $\frac{\partial \xi}{\partial x} = \frac{1}{J} \frac{\partial y}{\partial \eta}$ $\frac{\partial \eta}{\partial x} = -\frac{1}{J} \frac{\partial y}{\partial \xi}$ $\frac{\partial \xi}{\partial y} = -\frac{1}{J} \frac{\partial x}{\partial \eta}$ $\frac{\partial \eta}{\partial y} = \frac{1}{J} \frac{\partial x}{\partial \xi}$

- 2. What are the available structured grid generation techniques and explain the conformal mapping method. [16]
- 3. (a) What is Computational Fluid Dynamics? Explain with any two examples its applications in industrial manufacturing industry.
 - (b) Explain with diagrams the flow models that use finite control volume approach and infinitesimal fluid element approach. [8+8]
- 4. (a) Why computational fluid dynamics makes a distinction between conservation and non-conservation forms of governing equations? Explain with examples.
 - (b) Explain why conservation form of governing equations is important for calculations using shock capturing method with the help of an example of flow across a normal shock wave. [8+8]
- 5. (a) What is grid adaption
 - (b) Its need in a flow domain.
 - (c) Role played by it on the numerical solution with an example for two dimensional steady flow in a pipe. [4+4+8]
- 6. Prove that an expression for second-order central difference expression for the mixed derivative is $\left(\frac{\partial^2 u}{\partial x \partial y}\right)_{i,j} = \frac{u_{i+1,j+1} u_{i+1,j-1} u_{i-1,j+1} + u_{i-1,j-1}}{4\Delta x \Delta y} + O\left[\left(\Delta x\right)^2, \left(\Delta y\right)^2\right]$ can be obtained. [16]
- 7. (a) Derive the continuity equation $\partial \rho / \partial t + \nabla \bullet (\rho V) = 0$ assuming appropriate flow model.
 - (b) Derive the continuity equation $D\rho/Dt + \rho\nabla \bullet \nabla = 0$ assuming appropriate flow model. [8+8]
- 8. Classify the following partial differential equations according to their nature as elliptic/parabolic or hyperbolic

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Set No. 1

- (a) Unsteady Thermal Conduction Equation: $\partial T/\partial t = \alpha \partial^2 T/\partial t^2$
- (b) Laplace's Equation: $\partial^2 \phi / \partial x^2 + \partial^2 \phi / \partial y^2 = 0$
- (c) Second-order wave equation: $\partial^2 u/\partial t^2 = c^2 \partial^2 u/\partial x^2$
- (d) First order wave equation: $\partial u/\partial t + c\partial u/\partial x = 0.$ [4+4+4+4]

 $\mathbf{R05}$

Time: 3 hours

Set No. 3

III B.Tech II Semester Examinations, APRIL 2011 COMPUTATIONAL AERODYNAMICS Aeronautical Engineering

Max Marks: 80

Answer any FIVE Questions All Questions carry equal marks *****

- 1. (a) Derive the continuity equation $\partial \rho / \partial t + \nabla \bullet (\rho V) = 0$ assuming appropriate flow model.
 - (b) Derive the continuity equation $D\rho/Dt + \rho\nabla \bullet \nabla = 0$ assuming appropriate flow model. [8+8]
- 2. What are metrics and derive the relationship between the direct and inverse metrics. [16]

i.e.
$$\frac{\partial \xi}{\partial x} = \frac{1}{J} \quad \frac{\partial y}{\partial \eta} \qquad \frac{\partial \eta}{\partial x} = -\frac{1}{J} \quad \frac{\partial y}{\partial \xi}$$

 $\frac{\partial \xi}{\partial y} = -\frac{1}{J} \quad \frac{\partial x}{\partial \eta} \qquad \frac{\partial \eta}{\partial y} = \frac{1}{J} \quad \frac{\partial x}{\partial \xi}$

- 3. (a) What is Computational Fluid Dynamics? Explain with any two examples its applications in industrial manufacturing industry.
 - (b) Explain with diagrams the flow models that use finite control volume approach and infinitesimal fluid element approach. [8+8]
- 4. Prove that an expression for second-order central difference expression for the mixed derivative is $\left(\frac{\partial^2 u}{\partial x \partial y}\right)_{i,j} = \frac{u_{i+1,j+1} u_{i+1,j-1} u_{i-1,j+1} + u_{i-1,j-1}}{4\Delta x \Delta y} + O\left[\left(\Delta x\right)^2, \left(\Delta y\right)^2\right]$ can be obtained. [16]
- 5. What are the available structured grid generation techniques and explain the conformal mapping method. [16]
- 6. Classify the following partial differential equations according to their nature as elliptic/parabolic or hyperbolic
 - (a) Unsteady Thermal Conduction Equation: $\partial T/\partial t = \alpha \partial^2 T/\partial t^2$
 - (b) Laplace's Equation: $\partial^2 \phi / \partial x^2 + \partial^2 \phi / \partial y^2 = 0$
 - (c) Second-order wave equation: $\partial^2 u / \partial t^2 = c^2 \partial^2 u / \partial x^2$
 - (d) First order wave equation: $\partial u/\partial t + c\partial u/\partial x = 0.$ [4+4+4+4]
- 7. (a) Why computational fluid dynamics makes a distinction between conservation and non-conservation forms of governing equations? Explain with examples.
 - (b) Explain why conservation form of governing equations is important for calculations using shock capturing method with the help of an example of flow across a normal shock wave. [8+8]

R05

Set No. 3

- 8. (a) What is grid adaption
 - (b) Its need in a flow domain.
 - (c) Role played by it on the numerical solution with an example for two dimensional steady flow in a pipe. [4+4+8]